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# Discrete element method analysis of single wheel performance for a small lunar rover on sloped terrain

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## Abstract

The purpose of this study is to analyze the performance of a lugged wheel for a lunar micro rover on sloped terrain by a 2D discrete element method (DEM), which was initially developed for horizontal terrain. To confirm the applicability of DEM for sloped terrain locomotion, the relationships of slope angle with slip, wheel sinkage and wheel torque obtained by DEM, were compared with experimental results measured using a slope test bed consisting of a soil bin filled with lunar regolith simulant. **Among the lug parameters investigated, a lugged wheel with rim diameter of 250 mm, width of 100 mm, lug height of 10 mm, lug thickness of 5 mm, and total lug number of 18 was found, on average, to perform excellently in terms of metrics, such as slope angle for 20% slip, power number for self-propelled point, power number for 15-degree-slope and power number for 20% slip.** The estimation of wheel performance over sloped lunar terrain showed an increase in wheel slip, and the possibility exists that the selected lugged wheel will not be able to move up a slope steeper than 20 degrees.

**Key words:** Computational mechanics; Discrete element method; Soil-wheel system; Slope; Wheel performance; Lug; Lunar rover

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## 31 Nomenclature

32	$\mu$	friction coefficient between soil elements
33	$\mu^L$	friction coefficient between soil element and wheel (or lug) element
34	$\mu^w$	friction coefficient between soil element and wall element
35	$\omega$	angular velocity of wheel
36	$\phi$	angle of internal friction
37	$\rho$	density of soil element
38	$\rho_d$	dry bulk density
39	$\rho_e$	representative bulk density for damping coefficient calculation in DEM
40	$\rho_{max}$	maximum dry bulk density
41	$\rho_{min}$	minimum dry bulk density
42	$\theta$	inclination angle of slope
43	$\theta_{20\%}$	inclination angle of slope at 20% slip
44	$A_l$	total section area of lugs, such that $L_n BL_T$
45	$A_w$	area of possible contacting surface of wheel rim, such that $\pi BD_0$
46	$B$	width of wheel
47	$c$	cohesion
48	$C_l$	percent cover of total section area of lugs over area of wheel surface without
49		lugs, such that $C_l = A_l/A_w$
50	$C_n$	normal damping coefficient in DEM
51	$C_t$	tangential damping coefficient in DEM
52	$D$	outermost diameter of a wheel inclusive of lug height
53	$D_0$	rim diameter of a wheel
54	$D_r$	relative density of lunar regolith simulant
55	$f_x^+$	positive $x$ -component of contact reaction w. r. t. local axis
56	$f_x^-$	negative $x$ -component of contact reaction w. r. t. local axis
57	$H$	gross traction in slope locomotion, obtained as $H = \sum f_x^+$
58	$i$	slip of a wheel in slope locomotion, such that $i = 1 - (V_s/r_w\omega)$
59	$K_n$	normal spring constant between soil elements
60	$K_n^L$	normal spring constant between soil element and wheel (or lug) element
61	$K_n^w$	normal spring constant between soil element and wall element
62	$K_t$	tangential spring constant between soil elements
63	$K_t^L$	tangential spring constant between soil element and wheel (or lug) element
64	$K_t^w$	tangential spring constant between soil element and wall element
65	$L_H$	height of lug
66	$L_n$	total number of lug for a wheel
67	$L_T$	lug thickness
68	$m_e$	equivalent mass of soil element, defined as $\rho_e \times$ (volume of largest soil ele-
69		ment)
70	$P$	drawbar pull
71	$P_d$	net traction in slope locomotion, obtained as $P_d = H - R_r$
72	$PN$	power number, defined by $T\omega/WV_a$
73	$P/W$	pull coefficient, dimensionless, with respect to drawbar pull $P$ or $P_x$

- 74  $P_x$  drawbar pull w.r.t. local  $x$ -axis,  $P_x = W \sin \theta$
- 75  $R_r$  motion resistance in slope locomotion, obtained as  $R_r = \sum f_x^-$
- 76  $r_w$  rolling radius of a wheel, approximated to be an outermost radius of wheel
- 77 as  $r_w = D/2$
- 78  $T$  wheel torque, obtained as  $T = Hr_w$
- 79  $T/(Wr_w)$  torque coefficient
- 80  $V_s$  translational velocity of wheel along local  $x$ -axis
- 81  $V_X$  translational velocity of wheel along global  $X$ -axis
- 82  $V_Y$  translational velocity of wheel along global  $Y$ -axis
- 83  $W$  Vertical contact load of wheel w. r. t. global axis
- 84  $W_y$  normal contact load of wheel w. r. t. local axis,  $W_y = W \cos \theta$
- 85  $x$  abscissa of local coordinate system on a slope, positive for ascending direc-
- 86 tion
- 87  $X$  abscissa of global coordinate system, positive for rightward direction
- 88  $y$  ordinate of local coordinate system on a slope
- 89  $Y$  ordinate of global coordinate system, positive for upward direction
- 90  $z$  sinkage of wheel w. r. t. local  $y$ -axis
- 91  $Z$  sinkage of wheel w. r. t. global  $Y$ -axis
- 92  $z_a$  average sinkage of wheel w. r. t. local  $y$ -axis
- 93  $Z_e$  depth of soil w. r. t. global  $Y$ -axis

## 94 1 Introduction

95 The Selenological and Engineering Explorer (SELENE) was a Japanese lunar ex-  
96 ploration project that ran from December 2007 to June 2009. Using a remote sens-  
97 ing system on an observational orbiter, named “Kaguya,” SELENE obtained the  
98 first precise data on lunar surface geometry and mineral composition. A follow-up  
99 project, SELENE-2, is now under discussion at JAXA. Its main mission objectives  
100 would be to land on the Moon and obtain in situ geological samples using a small  
101 robotic rover [1]. To keep the payload of the H-II rocket to a minimum, the rover  
102 vehicle must be as small and light as possible. The envisaged maximum mass of  
103 the rover is 100 kg [2]. The rover can use either wheels or tracks for locomotion.

104 Since one of the candidate landing sites for SELENE-2 would be around the cen-  
105 tral peak of a crater, which should provide abundant geological information on the  
106 origin of the Moon, initial discussions on terramechanics for SELENE-2 had been  
107 focused on the mobility of a rover over the soft powdery lunar regolith accumu-  
108 lated over the peak. For this reason, an experimental investigation was started in an  
109 indoor horizontal soil bin with a lunar regolith simulant at Tsukuba Space Center  
110 (TSC) of NASDA (currently JAXA). Experiments with a rigid wheel resembling a  
111 conventional tire with no lugs, showed difficulty of locomotion even on horizontal  
112 terrain condition for a smaller drawbar load [3]. Moreover, developing a perfor-  
113 mance prediction model for such a 3D wheel shape proved difficult. Therefore, we



114 decided to develop simpler wheels with straight lugs, which may be approximated  
115 as 2D shapes, and a PC-based 2D performance prediction model for such wheels.  
116 For these reasons, our study focused on a lugged wheel.

117 There are only a few published reports on the interaction between wheels and plan-  
118 etary terrain such as lunar regolith. Freitag *et al.* [4] investigated several flexible  
119 wheels of different tread patterns for the Apollo Lunar Roving Vehicle (LRV) and  
120 reported that a uniform or near-uniform distribution of contact pressure was desir-  
121 able and they found no particular advantage in reducing the contact pressure below  
122 3.45 kPa for their tested flexible wheels. Moreover, the performances of several  
123 LRV wheels were reported by using a single-wheel dynamometer system and a  
124 wheel with 50% covered tread pattern showed slightly superior performance in Lu-  
125 nar Soil Simulant test [5]. Successively, two LRV wheels were tested to determine  
126 the influence of wheel speed, acceleration, travel direction, the presence of a fender,  
127 or wheel load [6]. LRV wheels were however much larger than those envisaged for  
128 our lunar micro rover. Their study of the wheel performance on a slope of proto-  
129 type rover models used three metrics: the pull coefficient, torque coefficient, and  
130 the power number [4–6].

131 Moreover, the experiences of manufacturing of candidate wheels for Apollo LRV  
132 were recently reported, and the decision processes for selection of LRV wheels  
133 were summarized in detail [7]. The computational model NWVPM (Nepean Wheeled  
134 Vehicle Performance Model) was used to predict the mobility of various wheels for  
135 LRV, and its predictions were accurate within the range of soil conditions published  
136 in some reports [8].

137 For small wheels, Richter and Hamacher [9] attempted to simulate the locomotion  
138 performance of microrovers on the Martian surface for the European Space Agency  
139 (ESA) by applying Bekker's formula. They constructed a 19.2 cm diameter lugged  
140 rigid wheel, designed for a mobile instrument deployment device vehicle with a  
141 mass of 8.6 kg, that applies a load of 7.92 N per wheel. Richter *et al.* further de-  
142 veloped a predictive wheel-soil interaction model for Mars rovers using Bekker's  
143 approach with a combination of the nonlinear slip-sinkage relationship and the con-  
144 tact area based modification of the shear deformation modulus  $K$  [10]. Their mod-  
145 ified model became sufficiently accurate to predict the performance of the Solar  
146 Powered Exploration Rover (SOLERO) and JPL Mars Exploration Rover (MER)  
147 wheels on DLR Mars Soil Simulant C (MSS-C) soil.

148 Recently, experimental approaches to wheel performance on sloped terrain have  
149 been developed using a sloped test bed with a lunar regolith simulant. A possi-  
150 ble star shaped wheel with specially arranged lugs was reported, but a detailed  
151 evaluation of power consumption was not included [11]. Moreover, a specially de-  
152 signed small elastic wheel demonstrated superior performance to the rigid lugged  
153 wheel [12].

154 The discrete, or distinct, element method (DEM), initially proposed by Cundall  
155 [13], has become popular as a computational tool for dynamics of particles or pow-  
156 ders in science and engineering. In the assembly of particles, there should be some  
157 effects of the potential forces from ambient particles other than the locally contact-  
158 ing. However, the present computational capability is still limited in considering all  
159 such forces. Thus, in DEM, the local contact reaction of two contacting elements is  
160 only considered.

161 In principle, DEM is based on the equation of motion, where all forces (such as  
162 contact reaction and body force) acting on the element of interest are added to the  
163 force term. After calculating all forces for all elements, the equations of motion are  
164 integrated to obtain subsequent velocities and displacements. The DEM solution  
165 is generally based on an explicit integration whose stability is conditional [13],  
166 which implies that the time step should be as small as possible. Moreover, for stable  
167 analysis, there exists the allowable range of time step in terms of the computational  
168 cost.

169 The normal contact reaction is calculated by contact models, such as the linear  
170 spring model [13] and Hertz contact model [14]. Similarly, both the linear spring  
171 model [13] and Mindlin-Deresiewicz contact model [15] are applied in calculating  
172 the tangential contact reaction. Viscous reaction forces are also assumed in both  
173 the normal and tangential directions. The shape of discrete elements can freely be  
174 defined, but simple shapes, such as a circle for 2D, or sphere for 3D, are most  
175 popular in terms of the detection of contact [13]. Other elemental shapes, such as  
176 an ellipse [16], polygon [17], or a clump of two or more circles [18] have also been  
177 applied in 2D DEM in the past.

178 Although a strict analysis of DEM based on the real element radius of soil particles  
179 might be ideal [19], at present, the radius of the element can not be of the same order  
180 as that in the target soil particles because of the computational cost. In this sense,  
181 the DEM element is a virtual element with a representative element radius that is  
182 larger than the real soil particles [20]. Therefore, the parameters used in the contact  
183 model of DEs should be determined or calibrated with comparative experiments  
184 using similar particle conditions.

185 Off-road wheel performance could be analyzed with sufficient accuracy [3, 18, 19,  
186 21–25] by DEM. Furthermore, the performance of a lugged wheel on a horizon-  
187 tal lunar terrain could be predicted by simply reducing the gravity from 1 to 1/6  
188 G, while holding other DEM parameters—such as spring constants and damping  
189 coefficients—constant [24, 25]. Using ellipsoid for soil elements, application of 3D  
190 DEM to the wheel of the MER under various gravities was recently reported [26].  
191 The results indicated that the wheel torque increased almost linearly with gravity.  
192 The authors' group verified the use of constant DEM parameters in low gravity con-  
193 ditions through analysis of sand pile formation under low gravity by performing a  
194 DEM simulation and airplane experiments [27]. Note that the analysis of DEM for

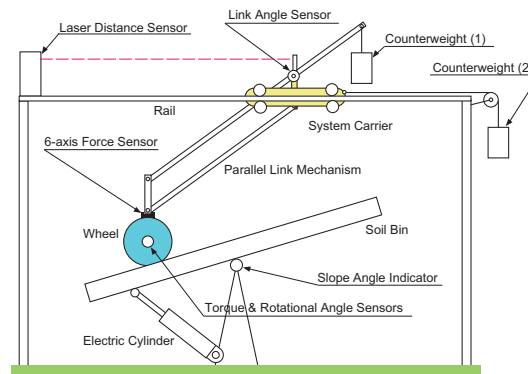


Fig. 1. Schematic diagram of the sloped soil bin and single wheel dynamometer

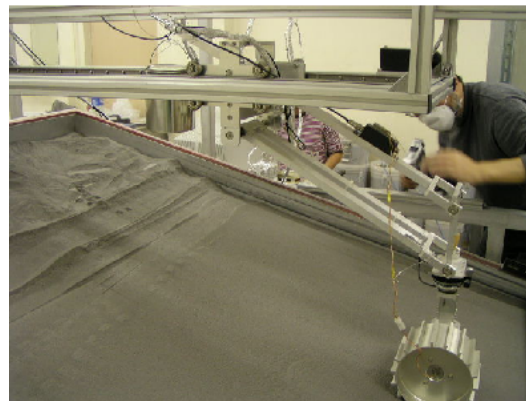


Fig. 2. View of the soil bin and single wheel dynamometer

195 sand pile formation under lunar gravity conditions was also reported but there was  
196 no comparison with experimental results [28].

197 The purpose of this study is to analyze the performance of a lugged wheel for a  
198 lunar micro rover on sloped terrain using the 2D DEM procedure previously de-  
199 veloped for horizontal terrain [24]. The analysis accuracy will be confirmed by  
200 experiments using a sloped test bed. The parameters for the lugged wheel are nu-  
201 merically investigated in terms of metrics, such as slope angle for 20% slip and  
202 power numbers, and a candidate wheel configuration is selected. Moreover, wheel  
203 performance on a sloped lunar terrain is predicted by DEM analysis with reduced  
204 relative gravitational acceleration using the selected lugged wheel configuration.

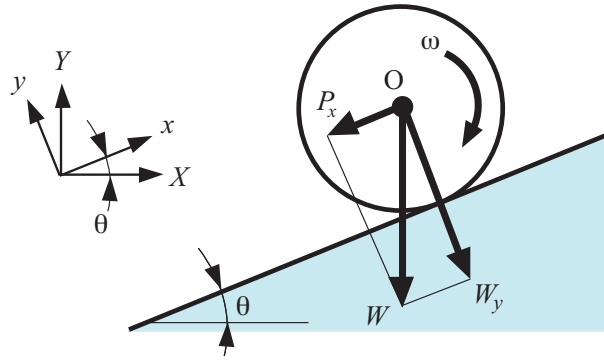


Fig. 3. Wheel travel on a slope

## 2 Experiments with single wheel performance on the slope

### 2.1 Sloped soil bin with a single wheel dynamometer

A sloped mobility test bed, consisting of a wide soil bin with effective inner dimensions of 1.5 m (width), 2.0 m (length), and 0.2 m (depth), which could be tilted, was constructed at Chofu Aerospace Center (CAC), JAXA [29]. The soil bin is axially rotated using a linear electrically driven actuator. A schematic configuration of the experimental facility is shown in Fig. 1.

A horizontal frame of the wheel carrier was constructed over the sloped soil bin. A target wheel attached at the end of a parallel link mechanism can freely sink vertically, with its weight controlled by the counterweight (1). Another counterweight (2) is adjusted to counteract the motion resistance of the system carrier so that the applied drawbar condition can be controlled properly. A six axis force sensor monitors the motion resistance by measuring the horizontal reaction. The vertical sinkage of a wheel is measured indirectly with a rotation angle sensor as the difference in the angle at the hinge point of the parallel link mechanism. Finally, a laser distance sensor monitors the horizontal travel distance of the wheel. Figure 2 shows a photo taken during the experiment.

Figure 3 shows the climbing wheel motion on sloped terrain with slope angle  $\theta$ . While we define the conventional global coordinate system  $(X, Y)$ , the local coordinate system of  $(x, y)$  is defined such that the  $x$ -axis is taken along the sloped surface as shown in Fig. 3.

In this simulation of wheel travel on a slope, the inputs are slope angle  $\theta$ , constant wheel rotation velocity  $\omega$ , and vertical contact load of the wheel  $W$ . In accordance with wheel dynamics over soil elements under contact, the wheel travels over the soil surface with the local travel velocity  $V_s$ , which can be calculated with respect

230 to a unit time step of numerical integration as

$$231 \quad V_s = V_X \cos \theta + V_Y \sin \theta \quad (1)$$

232 where  $V_X$  is the horizontal travel velocity,  $V_Y$  is the vertical velocity, and  $\theta$  is the  
233 slope angle. Note that  $V_X$  can be obtained from the distance signal of the laser  
234 distance sensor, and  $V_Y$  is calculated using the output from the angle sensor of the  
235 parallel link rotation in the experiment (see Fig. 1).

236 The slip  $i$  of a wheel can then be expressed using  $V_s$  such that

$$237 \quad i = 1 - \frac{V_s}{r_w \omega} \quad (2)$$

238 where  $r_w$  is the free rolling radius of the wheel, assumed as the outer radius of the  
239 wheel  $D/2$  including lug height  $L_H$ , since the measurement of free rolling radius is  
240 difficult for relatively soft soil conditions as in this study.

241 The pull coefficient is defined as the ratio of drawbar pull  $P_x$  to the vertical con-  
242 tact load of wheel  $W$ , i.e.,  $P_x/W$ . **As shown in Fig. 3, the pull coefficient is not a**  
243 **performance variable but a state variable, which can be expressed as  $P_x/W = \sin \theta$**   
244 **for sloped terrain locomotion.** The torque coefficient can be expressed as the ra-  
245 tio of wheel torque  $T$  to a product of the vertical contact load  $W$  and the rolling  
246 radius of the wheel  $r_w$ , i.e.,  $T/Wr_w$ . The power number  $PN$  measures the power  
247 consumption per unit of distance per unit of wheel weight in the locomotion from  
248 the starting point to the end point, and can be defined as follows [4]:

$$249 \quad PN = \frac{T\omega}{WV_s} \quad (3)$$

250 A wheel performs better if **the torque coefficient decreases** and if the power num-  
251 bers decrease.

## 252 2.2 Wheel specifications

253 Various test wheels made of aluminum were used in the experiments [23,24]. Their  
254 specifications are summarized in Table 1, where  $D$  is the total wheel diameter,  $D_0$   
255 is the rim diameter without lug,  $B$  is the width of the wheel,  $L_H$  is the height of the  
256 lugs,  $L_T$  is the thickness of the lugs, and  $L_n$  is the total number of lugs per wheel.  
257 The coefficient  $C_l$  represents the **percent cover** of the wheel lugs such that:

$$258 \quad C_l = \frac{A_l}{A_w} = \frac{L_n B L_T}{\pi B D_0} = \frac{L_n L_T}{\pi D_0} \times 100(\%) \quad (4)$$

Table 1  
Wheel specifications for slope locomotion

Wheel No.	$D$ (mm)	$D_0$ (mm)	$B$ (mm)	$L_H$ (mm)	$L_T$ (mm)	$L_n$	$C_l$ (%)
1	220	200	50	10	5	18	14.3
2 <sup>†</sup>	220	200	100	10	5	18	14.3
3	220	200	150	10	5	18	14.3
4	170	150	100	10	5	18	19.1
5	270	250	100	10	5	18	11.5
6	210	200	100	5	5	18	14.3
7	220	200	100	10	5	36	28.6
8	240	200	100	20	5	18	14.3
9	220	200	100	10	10	18	28.6

<sup>†</sup>Wheel No. 2 is the base condition of reference.

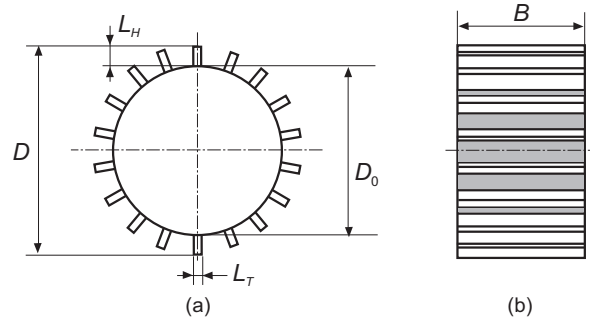


Fig. 4. Lug parameters

where  $A_w$  is the area of possible contacting surface of wheel rim, and  $A_l$  is the total section area of lugs .

The definition of lug parameters is shown in Fig. 4; Fig. 4(a) shows the wheel side view, and Fig. 4(b) shows the front view of a wheel with width  $B$ .

### 2.3 Experimental conditions

In the experiment, a lunar regolith simulant, FJS-1, prepared by Shimizu Corporation, was used to fill the soil bin to a depth of 10 cm, which was the same depth as in previous experiments on horizontal terrain [3,23]. Since the data of in situ observation in the Apollo 15 mission [30] indicated that there might be a relative density layer of very dense (hard) in lunar terrain around a depth of 10 cm or more, the depth of the simulant in the experiments was set to 10 cm, which assumes the existence of hardpan below 10 cm. The physical properties of FJS-1 are summarized

Table 2  
Physical property of lunar regolith simulant FJS-1

Specific gravity of particle	2.94
Minimum bulk density $\rho_{min}$ (g/cm <sup>3</sup> )	1.4
Maximum bulk density $\rho_{max}$ (g/cm <sup>3</sup> )	2.0
Cohesion $c$ (kPa)	2.55
Angle of internal friction $\phi$ (deg)	37.2

Table 3  
Experimental conditions

Vertical wheel load $W$ (N)	9.8, 14.7, 19.6
Slope angle $\theta$ (deg)	0, 10, 15, 20, 25
Angular velocity of wheel $\omega$ (rad/s)	0.1, 0.2, 0.3

in Table 2 [29].

Before each experiment, the simulant was manually raked and moved up and down within the soil bin and leveled with a leveling plate. The dry bulk density  $\rho_d$  in the experiments were monitored frequently by a vane shear device at three locations on the soil surface. The obtained shear torque was converted to  $\rho_d$  based on the calibration, which represented the value of  $\rho_d$  at a depth of 4 cm. Throughout the experiments, the average of  $\rho_d$  was found to be 1.48 g/cm<sup>3</sup>. The relative density  $D_r$  can then be calculated by

$$D_r = \frac{\rho_{max}}{\rho_d} \left( \frac{\rho_d - \rho_{min}}{\rho_{max} - \rho_{min}} \right) \times 100(\%) \quad (5)$$

From Eq. (5),  $D_r$  could be experimentally obtained as 17.4%. Note that this relative density condition is classified as “loose” [4, 30].

Shear strength parameters of FJS-1, such as cohesion and angle of internal friction, were obtained by conventional triaxial compression test with confining pressures similar to the reported measurement for JSC-1 [31]. Cohesion is thought to be related to inter particle attractive force, which becomes dominant for particles of smaller diameter [29].

Experiments were conducted using the conditions listed in Table 3. The vertical contact load contains the wheel weight. Experiments were repeated twice under the same conditions.

In the numerical simulation, the vertical contact load  $W$  was held constant for all slope angles, as was done in the experiment. The drawbar load  $P_x$  is applied as the slope angle  $\theta$  is changed, based on the relationship of  $P_x = W \sin \theta$ . The ratio of



Table 4  
Variation of ratio of  $P_x/W$  on the slope

$\theta$ (deg)	$P_x/W$
0	0.0
5	0.0872
10	0.1736
15	0.2588
20	0.3420
25	0.4226

$P_x/W(= \sin \theta)$  is summarized in Table 4. Acceleration forces are assumed to be negligible in the table.

### 3 DEM analysis of lugged wheel performance for sloped terrain

We previously reported the applicability of DEM for analyzing tractive performance of rigid lugged wheels for a lunar microrover on horizontal surfaces [23,24]. The analysis of wheel performance by DEM for various slopes is summarized in this section.

#### 3.1 Outline of analysis

Since the computational cost is still high in the application of 3D analysis, this study applied 2D DEM. The parameters for contact reaction are the spring constant, the viscous coefficient in normal and tangential directions, and the friction coefficient. This study uses a trial-and-error approach for an initial parameter guess, as in the previous study [24]. The justification of the selected parameter values is then verified by comparing the DEM with experimental results, as shown in Section 4.

Small DEM elements, embedded virtually at lug positions, are used for the contact check and contact reaction calculation between wheel lugs and soil [24]. All reactions from lug elements are added as the reaction of the wheel element.

Other small virtual DEM elements are embedded in three walls—two sides and one bottom—of the soil bin to facilitate the generation of sloped terrain conditions. These virtual wall elements are used only for contact calculation at the soil bin walls, and they do not move or rotate from contact reaction.



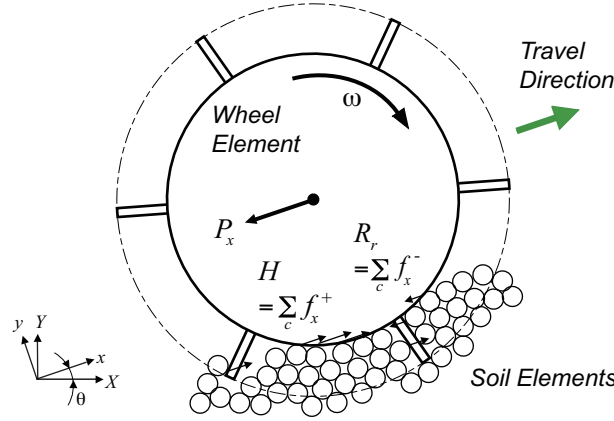


Fig. 5. Tractive effort and motion resistance on the slope for DEM

### 3.2 Wheel performance analysis on the slope

Each  $x$ -component of the contact reaction force  $f_x$  on a wheel and lug element is summed to obtain either the calculated gross traction  $H$  (if  $f_x \geq 0$ ), or motion resistance  $R_r$  (if  $f_x < 0$ ), with respect to the local  $x$ -axis as shown in Fig. 5.

After obtaining the gross traction  $H$  and the motion resistance  $R_r$  with DEM, we calculate the net traction  $P_d$  using  $P_d = H - R_r$  with respect to the local  $x$ -coordinate axis on the slope.

### 3.3 Preparation of DEM simulation

The parameters used in DEM analysis are summarized in Table 5. Note that the density of the regolith element of  $1.55 \text{ g/cm}^3$ , which corresponds to the value of the bulk density at about 5 cm below the lunar regolith surface [30], is used for calculating the damping coefficient.

The elemental density  $\rho \text{ (g/cm}^3\text{)}$  was calculated based on the initially generated depth of the regolith element  $Z_e$  using the following empirical relation from the Apollo program [30]:

$$\rho = 1.89 \times \frac{Z_e + 1.69}{Z_e + 2.9} \quad (6)$$

where  $Z_e$  is the depth of lunar regolith element (cm).

The spring constants ( $K_n$ ,  $K_t$ ) were initially set to the same values as in the previous study on wheel performance on horizontal terrain [23, 24]. The friction coefficient between soil DEs  $\mu$  is from the angle of internal friction of FJS-1. Friction coefficient of the soil bin walls  $\mu^w$  is set by assuming the composition of wall to be

Table 5  
Parameters in DEM simulation

Number of elements for regolith	6986
Diameter of the regolith element, <i>random</i> (mm)	2.8, 3.2, 3.6, 4.0
Regolith elemental density (g/cm <sup>3</sup> )	1.55 <sup>†</sup>
Mass of the wheel (g)	500
Diameter of the lug element (mm)	2.5
Diameter of the wall element (mm)	2.5
Slope angle $\theta$ (deg)	0, 5, 10, 15, 20, 25
Vertical contact load $W$ (N)	9.8, 14.7, 19.6
Drawbar load $P_x$ (N)	$W \sin \theta$
Angular velocity of the wheel $\omega$ (rad/s)	0.138
Duration of soil consolidation (s)	1.0, 6.0
Duration of vertical sinkage (s)	1.0
Simulation time for wheel travel (s)	40.0
Time step increment (s)	0.00005
Normal spring constants $K_n$ , $K_n^w$ and $K_n^L$ (N/m)	10,000
Tangential spring constants $K_t$ , $K_t^w$ and $K_t^L$ (N/m)	500
Friction coeff. between regolith elements $\mu$	0.75
Friction coeff. for wall contact $\mu^w$	0.75
Friction coeff. for wheel contact $\mu^L$	0.5

<sup>†</sup> This density value is used for damping coefficient calculation.

soil, while that for the wheel and lug contact  $\mu^L$  is the same as in the previous study. Damping coefficients were calculated using the critical damping formula, such that  $C_n = 2\sqrt{m_e K_n}$  for a normal damper and  $C_t = 2\sqrt{m_e K_t}$  for a tangential damper, where  $m_e$  is an equivalent mass representatively calculated by using  $\rho = 1.55 \text{ g/cm}^3$ ,  $B$  and  $r_{max}$ .

Moreover, in Table 2, the lunar regolith simulant FJS-1 shows cohesion as well as internal friction. As reported in our previous studies [23, 24], DEM results showed sufficient accuracy when compared with the experimental result, although the effect of cohesion was not explicitly introduced in the contact model in our DEM analysis. The introduction of cohesive reaction in a DEM contact model can easily be realized by an additional parallel tensile spring model [32], or an internal locking force model [33].

In the preliminary DEM analysis, the wheel rotational velocity did not exhibit a

significant effect on wheel performance parameters, such as slip, gross traction, and net traction. Thus, the data from the previous paper [24], of  $\omega = 0.138$  rad/s, is used in the analysis unless otherwise stated. Table 1 also lists the parameters for lug configuration and wheel diameter used in DEM.

DEM simulation for wheel performance is divided into four stages: (i) initial soil consolidation by their own weight of elements with the horizontal soil bin; (ii) rotation of the soil bin to the required slope angle; (iii) secondary consolidation with free wheel sinkage on to sloped soil surface and (iv) wheel travel simulation on sloped terrain [23].

In the first stage, the preparation of soil discrete elements is performed by analyzing the consolidation of soil DEs with their own weight of the elements from the initial regular configuration of DEs in the soil bin. The preparation of the soil condition in the experiment, however, involves manual mixing of the simulant with a hand rake, and leveling the disturbed surface by sliding a leveling blade over the edge of the soil bin. However, the present 2D DEM cannot include such preparation procedures due to computational cost as well as the limited degrees of freedom of the analysis.

At the wheel travel stage, the wheel sinkage was calculated as the local average sinkage  $z_a$  from the reference where the wheel begins to rotate. This avoids the difficulty of defining zero wheel sinkage with various lug conditions running over the uneven sloped surface of random soil DEs.

## 4 DEM analysis results and comparison with experimental results

### 4.1 Relationship of slope angle and slip

The specific wheel condition of  $D_0 = 200$  mm,  $B = 100$  mm, and  $W = 14.7$  N was selected for calibration of DEM analysis so that the selected DEM parameters could be verified sufficiently to obtain comparable results from experiments.

Figure 6(a) shows the result of a wheel with  $L_H = 10$  mm and  $L_n = 18$ . A large difference between DEM analysis and experiments can be seen at a slope of 15 deg, where the slip difference reaches 20%. Figure 6(b) indicates a smaller wheel slip with  $L_H = 20$  mm and the same  $L_n = 18$  at a slope of 25 deg in both DEM analysis and experiments. The largest slip difference between DEM and experiments is 16.3% at a slope of 20 deg. In Fig. 6(c), DEM and experimental results show similar behavior, but the difference in slip becomes large at the slope angles of 15 and 25 degrees in the experimental results. Comparing experimental results with numerical ones in Fig. 6, it can be stated that the selected parameters listed in Table 5 are sufficient for the DEM analysis of wheel performance on sloped terrain.

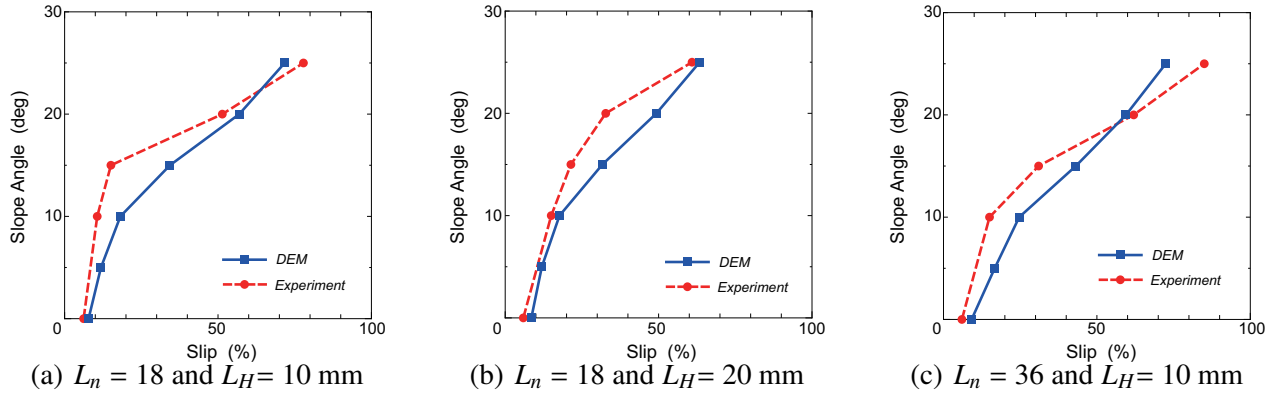


Fig. 6. Result of slip and slope (1)

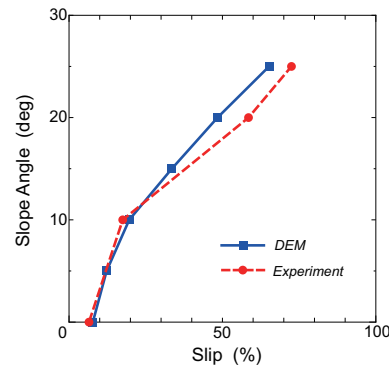


Fig. 7. Result of slip and slope (2)

To further verify the performance of DEM, a wheel with a larger vertical contact load of  $W = 19.6$  N with  $L_H = 20$  mm and  $L_n = 18$  was analyzed and the **result of DEM** was compared with recent experimental results [12], as shown in Fig. 7. The figure clearly indicates that the current DEM analysis can be applied with sufficient accuracy over a wide range of experimental conditions.

#### 4.2 Relationship of slope angle and wheel sinkage

Experimental results for the wheel with  $L_H = 20$  mm and  $L_n = 18$  for slopes of 10, 15, 20, and 25 deg are shown in Fig. 8. The straight solid lines indicate the ideal slope lines for the given caption: blue for 10 deg, green for 15 deg, purple for 20 deg, and red for 25 deg. It is clear that the wheel on a 25 deg slope travels with significant wheel slip and sinkage; the wheel travel is equivalent to that over the terrain with a slope of 20 deg. The DEM result (dashed lines) also indicates a similar reduction in wheel travel equivalence from 25 deg to 20 deg, as shown in Fig. 8. Other wheel conditions, such as  $L_H = 10$  mm and  $L_n = 36$ , showed a similar equivalence in the results of experiments and DEM to that seen in Fig. 8. Thus, a slope of 20–25 deg might be the maximum for wheel locomotion.

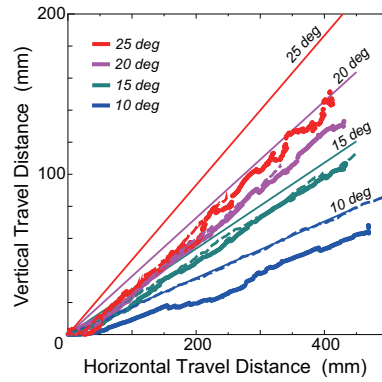


Fig. 8. Experimental and numerical wheel sinkage on sloped terrain

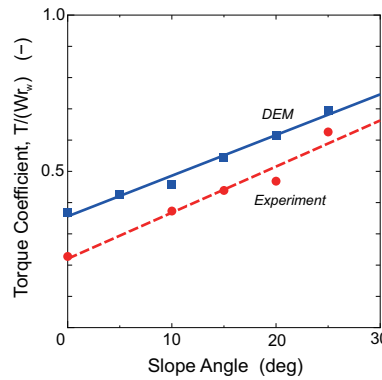


Fig. 9. Slope angle and wheel torque

### 399 4.3 Relationship of slope angle and wheel torque

400 Figure 9 shows the relationship between the slope angle and torque coefficient. The  
 401 wheel is  $D_0 = 200$  mm,  $L_H = 20$  mm,  $L_n = 18$ ,  $B = 100$  mm, and  $W = 14.7$  N. DEM  
 402 analysis obtains the wheel torque  $T$  as  $T = Hr_w$ , where  $H$  is the gross traction,  
 403 and  $r_w$  is the freely rolling radius of the wheel, which is assumed to be  $r_w = D/2$ .  
 404 The torque coefficient is then calculated as  $T/(Wr_w) = (Hr_w)/(Wr_w) = H/W$ . In  
 405 DEM, the wheel torque may be obtained directly from the wheel contact reaction.  
 406 However, the DEM program in this study shares a common subroutine for contact  
 407 reaction with that in our previous study [24]. Therefore, this study does not use  
 408 the wheel torque directly obtained at the wheel element. The torque coefficient  
 409 obtained by DEM analysis overestimates the experimental result by as much as of  
 410 0.15, although both show a similar linear increase with respect to the slope angle.

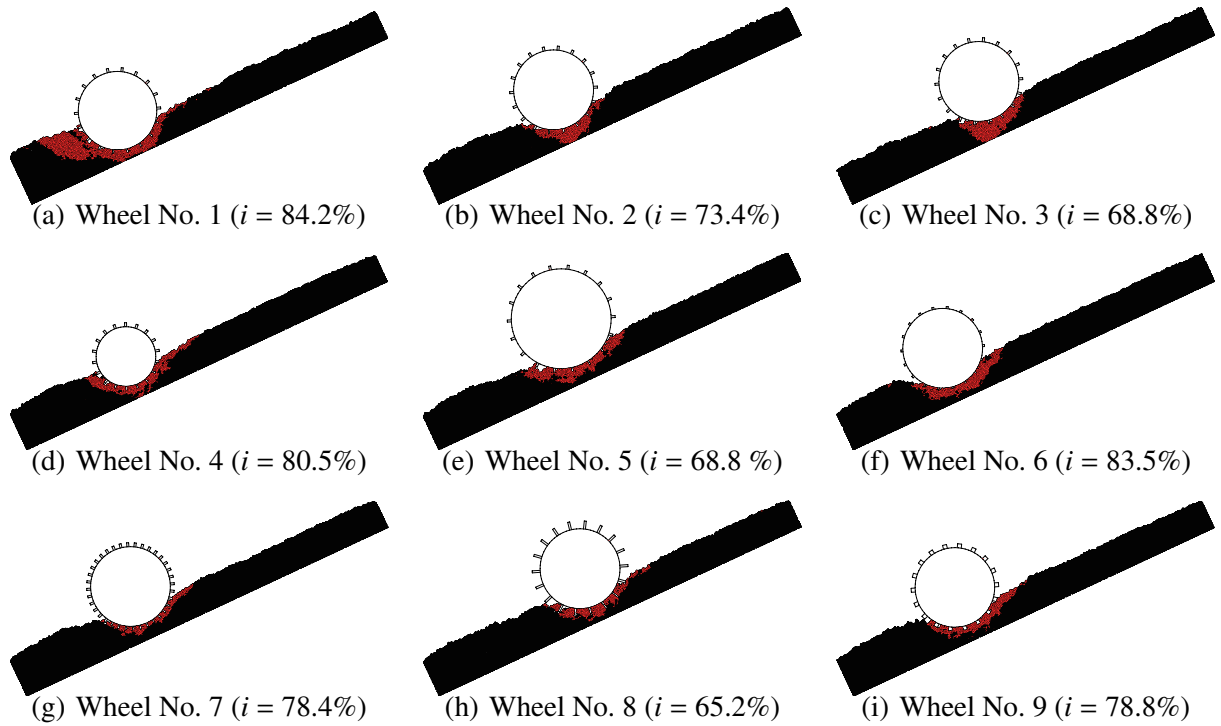


Fig. 10. Instantaneous soil flow under a wheel on 25-degree slope

## 5 Parametric investigation of lugged wheel performance on a sloped terrain by DEM

### 5.1 Remarks on soil flow under a wheel

Figure 10 shows the DEM result of instantaneous soil flow on 25-degree slope for each wheel in Table 1. The red region expresses the soil elements whose displacement becomes larger than 1 mm within a time step of 0.1 sec. Comparing the results for wheel numbers 1 to 3 [Figs. 10(a) to (c)] clearly shows that an increase in wheel width contributes to the prevention of sinkage. Moreover, the red region, indicating the area activated by lugs, clearly did not extend widely but remained close to the lug area in Figs. 10(g) to (i). Because of the effect of soil flow between or under the wheel lugs and the induced instability of soil with respect to internal friction angle, the figure shows that the soil element in front of the wheel flows by itself. Note that similar activated region with slip is seen in Figs 10(g) and 10(i) at the same coefficient of  $C_l = 28.6\%$  for  $D_0 = 200$  mm.

With respect to the bottom baseline of soil bin, an outline of mobilized zone of soil reaches to the bottom and is distorted in case of wheel number 1 because of the increased slip and sinkage of the wheel as seen in Fig. 10(a). Other wheels would not suffer significant effect of bottom wall of soil bin in terms of their outline shapes. For the effect of different diameter of wheel, wheel number 4 with small diameter

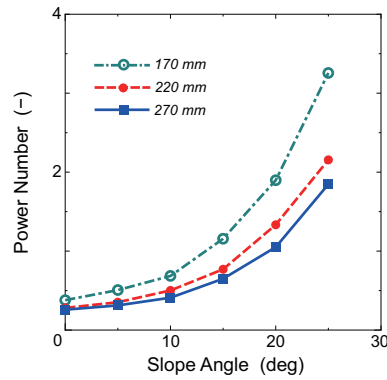


Fig. 11. Effect of the wheel diameter on the power number

[Fig. 10(d)] shows large sinkage, while wheel number 5 with large diameter [Fig. 10(e)] portrays small sinkage. The outline of mobilized zone under wheel number 5 expands widely, and that for wheel number 4 also grows wide under the wheel, because of increased slip and sinkage. On the other hand, wheel number 7 with  $L_n = 36$  [Fig. 10(g)] shows the smallest mobilized zone of soil in all wheel conditions, caused by the action of densely distributed lugs.

## 5.2 Effect of the wheel diameter on the power number

Figure 11 shows the effect of the wheel diameter  $D$  on the  $PN$ , with DEM result for wheels with  $D_0 = 150, 200$ , and  $250$  mm;  $L_H = 10$  mm;  $L_n = 18$ ;  $B = 100$  mm; and  $W = 19.6$  N. In DEM, the circumferential velocity at the end of the lug was kept constant at  $1.52$  cm/s by adjusting the angular velocity  $\omega$  to observe the geometrical effect of the wheel diameter.

For the largest difference in  $PN$  at a slope of  $25$  degrees, each DEM result is summarized as follows: the gross traction for cases of  $D_0 = 150, 200$ , and  $250$  mm is  $12.45, 11.25$ , and  $10.78$  N respectively. Similarly, the running resistance is  $4.15, 2.94$ , and  $2.48$  N respectively. Moreover, the average sinkage is  $2.04, 1.71$ , and  $1.53$  cm respectively. Thus, it is understood that, since a larger wheel diameter increases the contact area of the wheel under the same vertical contact load  $W$  and net traction  $P_d$ , wheel sinkage would be reduced, resulting in a smaller rolling resistance and slip at larger slope angles.

## 5.3 Effect of wheel load on the power number

The simulation result for a wheel when  $W = 9.8, 14.7$ , and  $19.6$  N;  $L_n = 18$ ;  $D_0 = 200$  mm;  $B = 100$  mm, is shown in Fig. 12. In Fig. 12(a),  $L_H = 10$  mm, and in Fig. 12(b),  $L_H = 20$  mm.

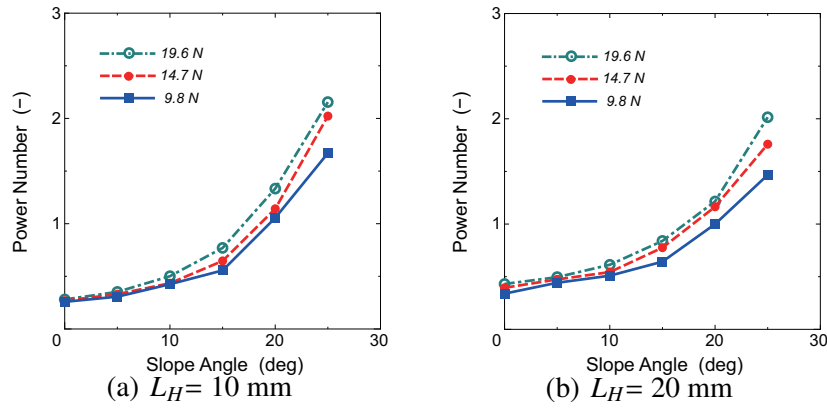


Fig. 12. Effect of wheel load on the power number

It is evident that the  $PN$  is reduced if the wheel load is low. For example, for the 25 degree slope shown in Fig. 12(a), the average sinkage  $z_a$  and the running resistance  $R_r$  were 1.18 cm and 1.46 N for  $W = 9.8$  N, 1.59 cm and 2.18 N for  $W = 14.7$  N, 1.71 cm and 2.94 N for  $W = 19.6$  N. However, the ratio of  $H/W$ , i.e. the torque coefficient as stated in 4.3, remained almost constant at 0.572–0.574. The low contact load  $W$  results in small wheel sinkage for a given slope, which contributes to the low running resistance  $R_r$ . The low running resistance implies low wheel slip, which reduces the  $PN$ . The difference in  $PN$  for 14.7 N and 19.6 N decreases at a slope of 20 deg. Moreover, at slope angles of 0 and 5 deg, the  $PN$  difference for various  $W$  also decreases.

#### 5.4 Effect of wheel width on power number

Figure 13 summarizes the simulation result for wheels with  $B = 50, 100$ , and  $150$  mm;  $L_H = 10$  mm;  $L_n = 18$ ;  $D_0 = 200$  mm; and  $W = 19.6$  N. The figure shows the effect of wheel width  $B$  on the  $PN$ . In all cases, if the wheel width increases, the  $PN$  decreases because of reduced wheel sinkage, reduced running resistance, and, therefore, of reduced wheel slip, as  $i = 84.2\%$  for a 50-mm-wide wheel and  $i = 68.8\%$  for a 150-mm-wide wheel. Note that the difference in  $PN$  between widths of 100 and 150 mm is not as large as the difference between 50 and 100 mm, when  $L_n = 18$  [Fig. 13(a) and (b)]. Among the three wheel conditions, a wheel with  $L_n = 18$ ,  $L_H = 10$  or 20 mm would be effective in terms of a low  $PN$ . From the figure, it is evident that the effect of wheel width may not be linear for some given slope angles and that an optimum wheel width may exist. The latter is due to increasing motion resistance from bulldozing for increased wheel width, offsetting the reduced sinkage.



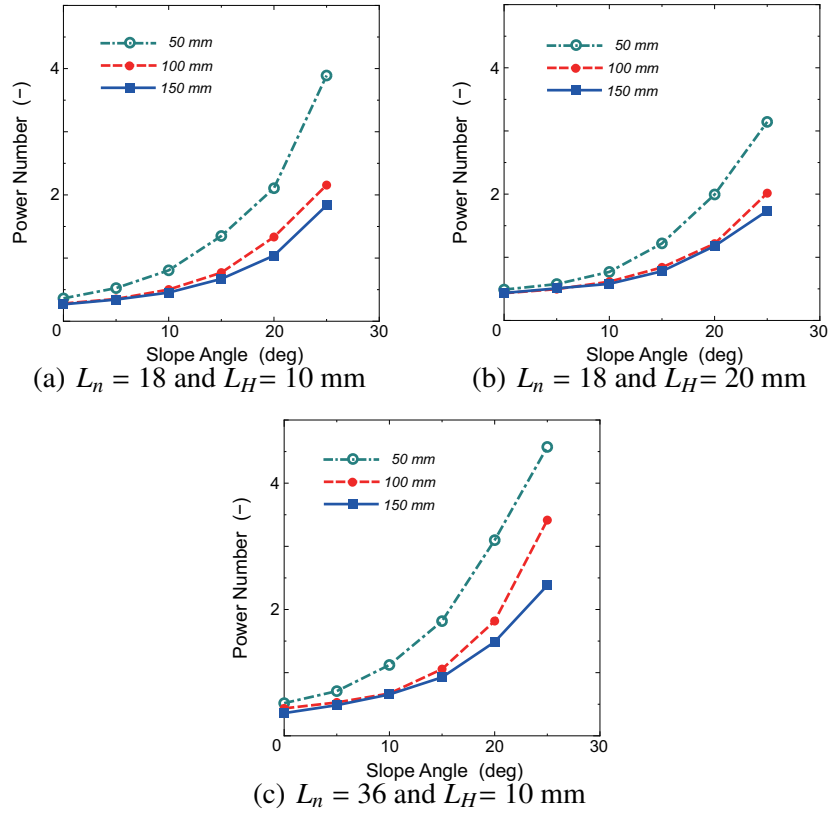


Fig. 13. Effect of wheel width on the power number

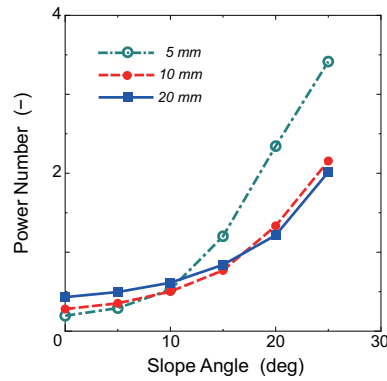


Fig. 14. Effect of lug height on the power number

#### 478 5.5 Effect of lug height on the power number

479 Figure 14 illustrates the effect of lug height on the  $PN$  for  $L_n = 18$ ,  $D_0 = 200$   
 480 mm, and  $W = 19.6$  N. The  $PN$  curve is lower for a lug height of  $L_H = 10$  or  
 481 20 mm, clearly showing that a higher lug height may be more effective for slope  
 482 locomotion.

483 In Fig. 14, it is noted that the curve for lug height of  $L_H = 20$  mm indicates largest  
 484 power number among three lug height conditions for smaller slips below or equal

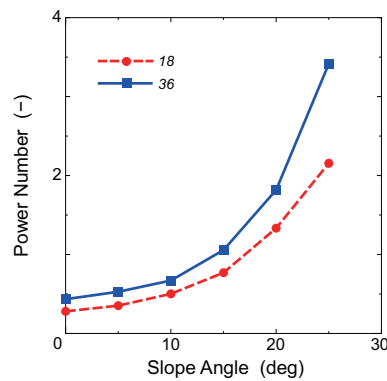


Fig. 15. Effect of lug number on the power number

485 to 10%. This would imply the possibility of trade-off between power consumption  
486 on horizontal terrain and that on sloped terrain for  $L_H = 20$  mm.

487 A wheel with  $L_H = 20$  mm [Fig. 10(h)] shows a narrow range of activated soil  
488 region that is mostly confined by the action of lugs within the outer most diameter  
489 of wheel  $D$  to create a clear line of shear. For wheels with smaller lug height [e.g.  
490 Fig. 10(f)], although the zone of mobilized soil extends outside of  $D$ , the soil around  
491 the wheel also becomes activated and the wheel itself does not generate larger gross  
492 traction through the action of its lugs because of the reduced friction coefficient of  
493 0.5 between the lug or wheel rim and soil elements; this increases wheel slip and  
494 thus the power number.

### 495 5.6 Effect of the total numbers of lugs on the power number

496 Figure 15 shows the simulation result for a wheel with  $L_n = 18$  and 36 while using  
497  $L_H = 10$  mm,  $D_0 = 200$  mm,  $W = 19.6$  N,  $L_T = 5$  mm, and  $B = 100$  mm. The effect  
498 of the number of lugs on the  $PN$  is clearly seen—the wheel with  $L_n = 18$  has a better  
499  $PN$  than that with  $L_n = 36$ .

500 To rationally describe the contributing mechanism of total number of lugs  $L_n$ , the  
501 coefficient  $C_l$  of different wheels is compared. From Table 1, the coefficient  $C_l$  for  
502 wheel number 2 is 14.3%, while that for wheel number 7 is 28.6%. In terms of the  
503 effect of the total number of lugs, a larger coefficient  $C_l$  means that the ratio of the  
504 soil part within the contact area of wheel becomes small. However, the lug faces,  
505 acting as a soil cutting tool, generate the gross traction in a lugged wheel. Moreover,  
506 the bottom lug face contributes to soil compaction that affects the wheel running  
507 resistance. From these conditions, the result of gross traction for wheel number 7  
508 becomes  $H = 14.45$  N while generating  $R_r = 6.14$  N as a negative effect. Thus, by  
509 cancelling the benefit of gross traction with running resistance under a small ratio  
510 of soil, the wheel with  $L_n = 36$  displayed a lower performance in terms of its  $PN$ .

511 Within the combination of investigated parameters, a larger wheel with fewer lugs

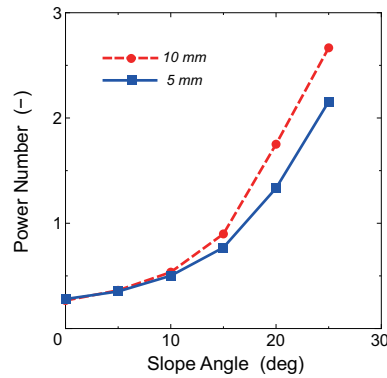


Fig. 16. Effect of the lug thickness on the power number

(e.g.,  $D_0 = 250$  mm and  $L_n = 18$ ) may be suitable for a lunar micro rover in terms of the total wheel mass. Note that when the wheel diameter increases, e.g., to 300 mm, the effect of the number of lugs on the  $PN$  may differ, since the number of lugs within the contact area under the wheel will be small for  $L_n = 18$ .

### 5.7 Effect of the lug thickness on the power number

Figure 16 shows the simulation result for a wheel with  $L_n = 18$ ,  $L_H = 10$  mm,  $D_0 = 200$  mm,  $W = 19.6$  N,  $B = 100$  mm, when the lug thickness is either  $L_T = 5$  or 10 mm. It is clear that the effect of thickness on the  $PN$  is almost the same for slope angles up to 10 degrees. For slope angles steeper than 10 degrees, a wheel with  $L_T = 5$  mm shows better performance in terms of the  $PN$ , but the difference in  $PN$  for a slope of 25 deg might be not so large, since there is only a slight difference in coefficient  $C_l$  and of wheel slip, which would contribute to the difference, and other factors, such as the average sinkage, gross traction, and running resistance, had almost the same values for two values of  $L_T$ . The coefficient  $C_l = 28.6\%$  for  $L_T = 10$  mm is the same as that for the wheel with  $L_n = 36$  (wheel number 7 in Table 1), and the wheel might become similar to a wheel with almost no lug effect. For such cases, the generation of gross traction in DEM may depend on the friction coefficient  $\mu^L$  between the wheel and soil rather than the friction coefficient  $\mu$  between soil DEs.

### 5.8 Remarks on lug parameters in relation with metrics

From the parametric observations in this section, the effect of the lug parameters are summarized in Table 6, using the representative metrics for wheel performance, such as  $\theta_{20\%}$ ,  $PN_{SP}$ ,  $PN_{15deg}$  and  $PN_{20\%}$  [4] for each wheel under  $W=19.6$  N.

For slope angle at 20% slip ( $\theta_{20\%}$ ), the wheel performance is regarded as better if its angle increases. For other metrics, the wheel has better power consumption, if

Table 6  
Representative metrics with superiority weight

Wheel No.	$\theta_{20\%}$ (deg)		$PN_{SP}$		$PN_{15deg}$		$PN_{20\%}$		Sum of weights
1	0.49	8	0.363	6	1.350	9	0.376	2	(25)
2	7.10	3	0.280	5	0.790	3	0.414	4	15
3	10.0	1	0.268	4	0.670	2	0.453	6	(13)
4	3.70	6	0.379	7	1.156	7	0.475	7	(27)
5	9.35	2	0.258	2	0.650	1	0.396	5	10
6	2.72	7	0.195	1	1.203	8	0.246	1	(17)
7	5.87	5	0.434	9	1.056	6	0.554	8	(28)
8	10.0	1	0.431	8	0.840	4	0.620	9	(22)
9	6.30	4	0.266	3	0.899	5	0.391	3	15
Average	6.17		0.319		0.955		0.436		

the  $PN$  decreases. An order of superiority is also shown by a simple integer for each metric, where a smaller number is input for higher performance. The last column in Table 6 summarizes the sum of four numbers, which implies the lowest sum might be regarded as the best wheel in terms of the four corresponding metrics. The sum with brackets indicates that some metrics in the table are out of the preferred range.

Comparing the metrics with respect to the corresponding average values shows that wheel numbers 2, 5 and 9 may be candidates for lugged wheels for the small lunar rovers under study. Wheel numbers 2 and 5 have similar lug parameters with different wheel diameters;  $D_0 = 20$  cm for the wheel number 2 and  $D_0 = 25$  cm for number 5. In terms of the smallest sum of weights, wheel number 5 may be a candidate wheel within the lug parameters investigated in this section. The selection of wheel number 5 may be the same result as in our previous DEM analysis on horizontal terrain [24]. Note that the diameter  $D$  of the wheel should be determined from other requirements, such as stowage of the rover vehicle in the lander.

The performance from DEM of a lugged wheel on a slope of 25 degrees, the result of tractive performance, such as gross traction  $H$ , net traction  $P_d$ , running resistance  $R_r$  and average sinkage  $z_a$ , which was integrated from the beginning of wheel rotation and end of wheel travel, is summarized in Table 7. The table verifies that wheel number 5 exhibits the best traction performance under the lug conditions examined, in terms of small running resistance and average sinkage. Wheel number 8 ( $L_H = 20$  mm and  $L_n = 18$ ) and wheel number 7 ( $L_H = 10$  mm and  $L_n = 36$ ) not only increased the gross traction but also increased running resistance while indicating the same average level of wheel sinkage.

It is noted that the effect of cohesion, which is not considered in the present DEM

Table 7  
Performance of lugged wheel on 25-degree slope

Wheel No.	$i(\%)$	$H(\text{N})$	$P_d(\text{N})$	$R_r(\text{N})$	$z_a(\text{cm})$
1	84.2	12.04	8.31	3.73	2.47
2	73.4	11.25	8.31	2.94	1.71
3	68.8	11.23	8.31	2.92	1.57
4	80.5	12.45	8.30	4.15	2.04
5	68.8	10.78	8.30	2.48	1.53
6	83.5	11.06	8.29	2.77	1.67
7	78.4	14.45	8.31	6.14	1.65
8	65.2	13.74	8.30	5.44	1.65
9	78.8	11.10	8.29	2.81	1.69

analysis, may become important in comparison of tractive performance of wheels with different contact areas as observed by  $z_a$ .

## 6 Prediction of lugged wheel performance on sloped lunar terrain

### 6.1 Introduction

The locomotion over lunar terrain has been a challenging topic for vehicle engineers ever since the US Apollo mission and the Luna mission of the USSR. The robotic rover Lunokhod 2 sank significantly in the vicinity of some craters, and the Apollo 15 LRV encountered isolated soft soil and became stuck [30]. Thus, predictions of wheel performance and the prospective rover design should take into account maneuverability in soft soil conditions.

The performance of a wheel on sloped lunar soft terrain is simulated using the selected wheel parameters ( $D_0 = 250$  mm;  $B = 100$  mm;  $L_H = 10$  mm;  $L_T = 5$  mm;  $L_n = 18$ ) under a vertical contact load of  $W = 14.7$  N. To observe the soil condition, the same contact load of  $W$  is used for both lunar terrain and earth terrain, which means the drawbar load  $P_x$  also becomes the same on the Earth and on the Moon depending on the slope angle.

Since adopting the same wheel contact load both for the Moon and Earth is expected to produce greater wheel sinkage, the initially generated thickness of soil DEs is increased to 15 cm, where the total number of soil DEs is 9980. Moreover, the length of the soil bin was extended, and for sloped terrain analysis the wheel was positioned in the middle of the soil bin to avoid the disturbance of soil flow

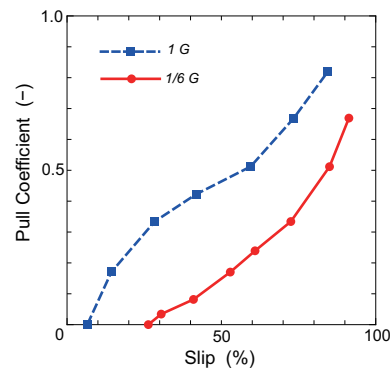


Fig. 17. Pull coefficient on horizontal terrain

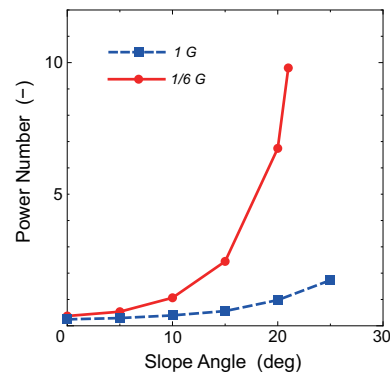


Fig. 18. Effect of gravitational acceleration on the power number

caused by its side walls. This produced a total of 15980 soil DEs for sloped lunar terrain analysis.

## 6.2 Pull coefficient obtained from the horizontal terrain condition

Firstly, the performance of wheel was obtained on the horizontal terrain. Various drawbar loads were input to simulate tractive performance of the wheel in 2D DEM [24]. The pull coefficient  $P/W$  at 1/6 G is shown in Fig. 17 along with the result at 1 G. From the figure, it is clear that the weak, or soft, condition of soil on the lunar gravity results in the increased sinkage of wheel and, thus, decreased output of pull coefficient. The slip of 26.3% for a zero pull coefficient at 1/6 G indicates reduced wheel mobility on horizontal lunar terrain.

## 6.3 Effect of gravitational acceleration on the power number

Figure 18 shows the  $PN$  under the effect of gravitational acceleration for slope locomotion. Note that the wheel cannot climb a slope of 25 degrees at 1/6 G. It is clear that the reduced gravity would increase the  $PN$  at all slope angles. The

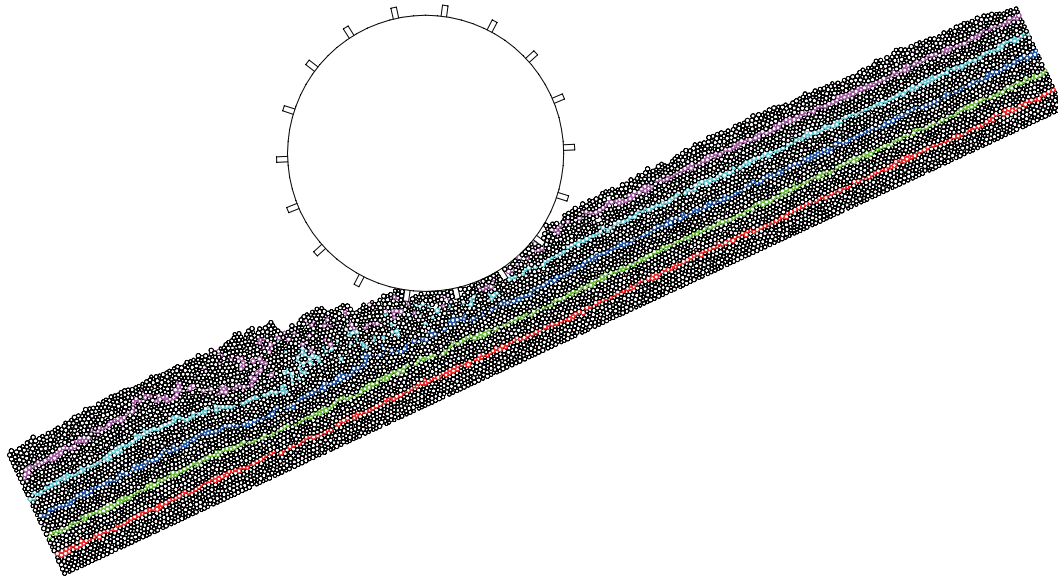


Fig. 19. Soil deformation under a wheel on a Earth slope of 25 degrees

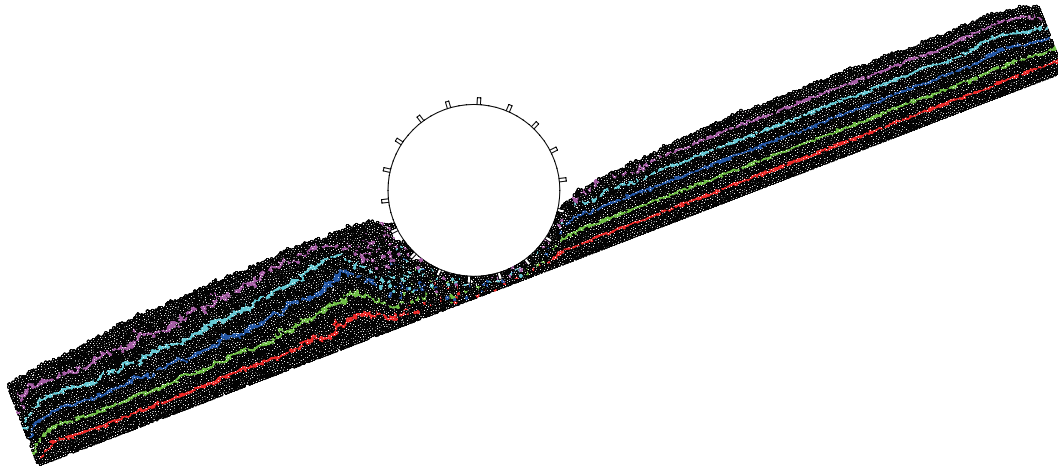


Fig. 20. Soil deformation under a wheel on a lunar slope of 21 degrees with extended soil bin condition

596 increased  $PN$ , especially for larger slope angles in the figure, implies an excessive  
597 energy expenditure due to the reduction of soil trafficability caused by reduced  
598 gravitational acceleration on the Moon.

#### 599 6.4 Comparison of soil deformation under the wheel

600 Figure 19 shows the results of soil deformation for Earth gravity at a slope angle of  
601 25 deg, where the wheel slip was 68.3%. The result of soil deformation for lunar  
602 gravity at a slope angle of 21 deg is shown in Fig. 20. In the figure, the slip was  
603 94.4% at 1/6 G, and the wheel could not climb the slope angle at 25 deg as it did at  
604 1 G. In both figures, the vertical contact load condition is the same.

The wheel sinks considerably with  $z_a = 3.01$  cm at  $1/6$  G because of the increased slip, since the weight of the soil element would be reduced by  $1/6$ , thereby reducing the compaction by its own weight even in the extended period of soil consolidation, as can be inferred from Fig. 20.

## 6.5 Discussion

It becomes clear that the candidate lugged wheel with  $D_0 = 250$  mm,  $B = 100$  mm,  $L_H = 10$  mm,  $L_T = 5$  mm,  $L_n = 18$  under  $W = 14.7$  N might have difficulty climbing a lunar terrain sloped more than 20 degrees. This is mainly due to the reduced trafficability of soil in a lunar environment.

Moreover, the preparation of the initial soil condition is an issue in application of the DEM. In the previous study [4], the following items were proposed for the experiments with simulants. Important parameters were relative density, cohesion, angle of internal friction, and penetration resistance gradient. Cohesion and the angle of internal friction can be regarded as input parameters in DEM, but the effect of cohesion is not yet implemented in our present analysis. The relative density and the penetration resistance gradient can be obtained as a result of the analysis, although our present DEM does not include these output.

The consolidation in the first stage of DEM analysis was thought to be a governing factor of soil conditioning on bulk density in low gravity conditions [27]. An inverse proportional factor of six was multiplied to the consolidation time of earth conditions to increase the duration of consolidation for lunar gravity of  $1/6$  G so that the soil DEs could be stabilized sufficiently by their own weight during consolidation. In the present analysis, however, the effect of extended time for lunar gravity could not be found because the initial generation of soil DEs was done in a rather dense configuration, where the fall height of each soil element could not be large. For the precise prediction of lugged wheel performance on the Moon, the preparation of initial condition of soil DEs should be investigated.

## 7 Conclusions

To confirm the applicability of DEM for sloped terrain locomotion, the relationships of slope angle with slip, wheel sinkage and wheel torque obtained by DEM, were compared with experimental results measured using a slope test bed consisting of a soil bin filled with lunar regolith simulant. Among the lug parameters investigated, a lugged wheel with  $D_0 = 250$  mm,  $B = 100$  mm,  $L_H = 10$  mm,  $L_T = 5$  mm,  $L_n = 18$  was found, on average, to have excellent performance in terms of metrics, such as  $\theta_{20\%}$ ,  $PN_{SP}$ ,  $PN_{15deg}$  and  $PN_{20\%}$ . The estimation of wheel perfor-



640 mance over sloped lunar terrain showed an increase in wheel slip, and a possibility  
641 that the selected lugged wheel might not have locomotion on a slope greater than  
642 20 degrees.

643 The low gravity on the Moon resulted in reduction of frictional forces between soil  
644 particles, observed as the increase in sinkage of wheel. Consequently, the effect of  
645 cohesion would become more significant in lunar gravity. An investigation into the  
646 relative influence of cohesion both at 1 G and at 1/6 G using DEM is recommended.  
647 Moreover, successive study on metrics for all wheel candidates in lunar gravity by  
648 DEM is suggested to verify the effect of gravity on wheel parameters decided in  
649 this analysis.

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## 655 References

- 656 [1] JAXA. JAXA  
657 Vision–JAXA 2025–. [http://www.jaxa.jp/about/2025/index\\_e.html](http://www.jaxa.jp/about/2025/index_e.html), accessed  
658 on March 8, 2008.
- 659 [2] Nishida S, Wakabayashi S. A study on mobility platform of lunar work rover. In:  
660 Preprints 26th Int. Symposium on Space Technology and Science, Hamamatsu, 2008:  
661 2008-k-57. (CD-ROM)
- 662 [3] Fujii H. DEM analysis for a wheel of lunar rover and regolith system interaction.  
663 Unpublished Bachelor of Agric. Sci. Thesis, Faculty of Agriculture, Kyoto University,  
664 Kyoto, 2001. (in Japanese)
- 665 [4] Freitag DR, Green AJ, Melzer KJ, Costes NC. Wheels for lunar vehicles. Journal of  
666 Terramechanics 1972; 8(3): 89–105.
- 667 [5] Green AJ, Melzer KJ. Performance of the Boeing LRV wheels in a lunar soil  
668 stimulant—Report 1 Effect of wheel design and soil. Technical Report M-71-10,  
669 Mobility and Environmental Division, U. S. Army Engineer Waterways Experiment  
670 Station, Vicksburg, Mississippi, 1971.
- 671 [6] Melzer, KJ. Performance of the Boeing LRV wheels in a lunar soil stimulant—  
672 Report 2 Effects of speed, wheel load, and soil. Technical Report M-71-10, Mobility

- 673 and Environmental Division, U. S. Army Engineer Waterways Experiment Station,  
674 Vicksburg, Mississippi, 1971.
- 675 [7] Asnani V, Delap D, Creager C. The development of wheels for the Lunar Roving  
676 Vehicle. *Journal of Terramechanics* 2009; 46: 89–103.
- 677 [8] Wong JY, Asnani VM. Study of the correlation between the performances of lunar  
678 vehicle wheels predicted by the Nepean wheeled vehicle performance model and test  
679 data. *Proc. IMechE, Part D: Journal of Automobile Engineering* 2008; 222: 1939–1954.
- 680 [9] Richter L, Hamacher H. Investigation the locomotion performance of planetary  
681 microrovers with small wheel diameters and small wheel loads. In: *Proc. 13th Int. Conf.*  
682 *ISTVS, Munich, 1999: 719–726.*
- 683 [10] Richter L, Ellery A, Gao Y, Michaud S, Schmitz N, Weiss S. A predictive wheel-soil  
684 interaction model for planetary rovers validated in testbeds and against MER Mars rover  
685 performance data. In: *Proc. 10th European Conf. ISTVS, Budapest, 2006: 1–27.*
- 686 [11] Iizuka K, Sato Y, Kuroda Y, Kubota T. Study on traversability with consideration of  
687 wheeled forms for lunar and planetary exploration rovers. *Trans. JSME* 2006; 72(724):  
688 148–153. (in Japanese)
- 689 [12] Wakabayashi S, Kohno Y, Nishida S. Evaluation of low-pressure mobility system for  
690 lunar vehicle. In: *Preprints 26th Int. Symposium on Space Technology and Science,*  
691 *Hamamatsu, 2008: 2008-k-56. (CD-ROM)*
- 692 [13] Cundall PA, Strack ODL. Discrete numerical model for granular assemblies.  
693 *Géotechnique* 1979; 29(1): 47–65.
- 694 [14] Johnson KL. *Contact mechanics.* Cambridge University Press, Cambridge, 1985.
- 695 [15] Mindlin RD, Deresiewicz H. Elastic spheres in contact under varying oblique forces.  
696 *Journal of Applied Mechanics* 1953; 20: 327–344.
- 697 [16] Sakaguchi E, Kawakami S, Tamura S, Tobita F. Simulation on discharging phenomena  
698 of grains by distinct element method–Influence of shapes of element on flowing states.  
699 *Journal of The Japanese Society of Agricultural Machinery* 1996; 58(4): 9–17. (in  
700 Japanese)
- 701 [17] Feng YT, Owen DRJ. An energy based corner to contact algorithm. *Discrete Element*  
702 *Methods* (Edited by Cook BK, Jensen RP), ASCE, 32–37, 2002.
- 703 [18] Asaf Z, Shmulevich I, Rubinstein D. Predicting soil-rigid wheel performance using  
704 distinct element methods. *Transactions of the ASABE* 2006; 49(3): 607–616.
- 705 [19] Horner DA, Peters JF, Carrillo A. Large scale discrete element modelling of vehicle-  
706 soil interaction. *Journal of Engineering Mechanics* 2001; 127(10): 10271032.
- 707 [20] Nakashima H, Konishi T, Toki Y. Numerical Analysis for Cone Penetration of  
708 Mesoscopic Soil Model. In: *Proc. of the Joint North America, Asia-Pacific ISTVS*  
709 *Conference and Annual Meeting of Japanese Society for Terramechanics, Fairbanks,*  
710 *June 23–26, 2007: 2007-31-0319, 1–9. (CD-ROM)*

- 711 [21] Oida A, Ohkubo S, Schwanghart H. Effect of tire lug cross section on tire performance  
712 simulated by distinct element method. In: Proc. 13th Int. Conf. ISTVS, Munich, 1999:  
713 345–352.
- 714 [22] Oida A, Ohkubo S. Application of DEM to simulate interaction between soil and tire  
715 lug. *Agricultural & Biosystems Engineering* 2000; 1(1): 1–6.
- 716 [23] Fujii H. An analysis of lugged wheel performance by DEM for the development  
717 of wheel for a lunar rover. Unpublished Master of Agric. Sci. Thesis, Division  
718 of Environmental Science & Technology, Graduate School of Agriculture, Kyoto  
719 University, Kyoto, 2003. (in Japanese)
- 720 [24] Nakashima H, Fujii H, Oida A, Momozu M, Kawase Y, Kanamori H, Aoki S,  
721 Yokoyama T. Parametric analysis of lugged wheel performance for a lunar microrover  
722 by means of DEM. *Journal of Terramechanics* 2007; 44(2): 153–162.
- 723 [25] Li JQ, Zou M, Jia Y, Ma WZ, Ren LQ, Li YW. Research the interaction between  
724 the rigid wheel and the lunar soil by DEM. In: Proc. of the Joint North America, Asia-  
725 Pacific ISTVS Conference and Annual Meeting of Japanese Society for Terramechanics,  
726 Fairbanks, June 23–26, 2007: 1–12. (CD-ROM)
- 727 [26] Hopkins MA, Johnson JB, Sullivan R. Discrete element modeling of a rover wheel  
728 in granular material under the influence of Earth, Mars, and Lunar Gravity. In: Proc.  
729 Earth & Space 2008, 11th ASCE Aerospace Division International Conference on  
730 Engineering, Science, Construction, and Operations in Challenging Environments, Long  
731 Beach, March 3–5, 2008: 1–7. (CD-ROM)
- 732 [27] Nakashima H, Shioji Y, Kobayashi T, Aoki S. Numerical analysis of sand flow  
733 under low gravity condition. In: Proc. of the Joint North America, Asia-Pacific ISTVS  
734 Conference and Annual Meeting of Japanese Society for Terramechanics, Fairbanks,  
735 June 23–26, 2007: ISTVS-2007-62-0147, 1-10. (CD-ROM)
- 736 [28] Ji S, Shen HH. Contrasting terrestrial and lunar gravity: angle of repose and incline  
737 flows, In: Proceedings of 10th ASCE Aerospace Division International Conference on  
738 Engineering, Construction, and Operations in Challenging Environments, 2006: 1–8.  
739 (CD-ROM)
- 740 [29] Wakabayashi S, Matsumoto K. Development of slope mobility testbed using lunar  
741 soil. JAXA Research and Development Memorandum, JAXA-RM-05-003, 2006: 1–11.  
742 (in Japanese)
- 743 [30] Heiken GH, Vaniman DT, French BM, Ed. Lunar sourcebook. Cambridge University  
744 Press, Cambridge, 1991.
- 745 [31] Willman BM, Boles WW, McKay DS, Allen CC. Properties of lunar soil simulant  
746 JSC-1. *Journal of Aerospace Engineering, ASCE*, 1995; 8(2): 77–87.
- 747 [32] Momozu M, Oida A, Yamazaki M, Koolen AJ. Simulation of a soil loosening process  
748 by means of the modified distinct element method. *Journal of Terramechanics* 2003; 39:  
749 207–220.

- 750 [33] Bui HH, Kobayashi T, Fukagawa R, Wells JC. Numerical and experimental studies of  
751 gravity effect on the mechanism of lunar excavations. *Journal of Terramechanics* 2009;  
752 46: 115–124.